

Presidential Documents

Title 3—

Executive Order 12803 of April 30, 1992

The President

Infrastructure Privatization

By the authority vested in me as President by the Constitution and the laws of the United States of America, and in order to ensure that the United States achieves the most beneficial economic use of its resources, it is hereby ordered as follows:

Section 1. *Definitions.* For purposes of this order:

(a) "Privatization" means the disposition or transfer of an infrastructure asset, such as by sale or by long-term lease, from a State or local government to a private party.

(b) "Infrastructure asset" means any asset financed in whole or in part by the Federal Government and needed for the functioning of the economy. Examples of such assets include, but are not limited to: roads, tunnels, bridges, electricity supply facilities, mass transit, rail transportation, airports, ports, waterways, water supply facilities, recycling and wastewater treatment facilities, solid waste disposal facilities, housing, schools, prisons, and hospitals.

(c) "Originally authorized purposes" means the general objectives of the original grant program; however, the term is not intended to include every condition required for a grantee to have obtained the original grant.

(d) "Transfer price" means: (i) the amount paid or to be paid by a private party for an infrastructure asset, if the asset is transferred as a result of competitive bidding; or (ii) the appraised value of an infrastructure asset, as determined by the head of the executive department or agency and the Director of the Office of Management and Budget, if the asset is not transferred as a result of competitive bidding.

(e) "State and local governments" means the government of any State of the United States, the District of Columbia, any commonwealth, territory, or possession of the United States, and any county, municipality, city, town, township, local public authority, school district, special district, intrastate district, regional or interstate governmental entity, council of governments, and any agency or instrumentality of a local government, and any federally recognized Indian Tribe.

Sec. 2. *Fundamental Principles.* Executive departments and agencies shall be guided by the following objectives and principles: (a) Adequate and well-maintained infrastructure is critical to economic growth. Consistent with the principles of federalism enumerated in Executive Order No. 12612, and in order to allow the private sector to provide for infrastructure modernization and expansion, State and local governments should have greater freedom to privatize infrastructure assets.

(b) Private enterprise and competitively driven improvements are the foundation of our Nation's economy and economic growth. Federal financing of infrastructure assets should not act as a barrier to the achievement of economic efficiencies through additional private market financing or competitive practices, or both.

(c) State and local governments are in the best position to assess and respond to local needs. State and local governments should, subject to assuring continued compliance with Federal requirements that public use be on reasonable and nondiscriminatory terms, have maximum possible freedom to

make decisions concerning the maintenance and disposition of their federally financed infrastructure assets.

(d) User fees are generally more efficient than general taxes as a means to support infrastructure assets. Privatization transactions should be structured so as not to result in unreasonable increases in charges to users.

Sec. 3. Privatization Initiative. To the extent permitted by law, the head of each executive department and agency shall undertake the following actions:

(a) Review those procedures affecting the management and disposition of federally financed infrastructure assets owned by State and local governments and modify those procedures to encourage appropriate privatization of such assets consistent with this order;

(b) Assist State and local governments in their efforts to advance the objectives of this order; and

(c) Approve State and local governments' requests to privatize infrastructure assets, consistent with the criteria in section 4 of this order and, where necessary, grant exceptions to the disposition requirements of the "Uniform Administrative Requirements for Grants and Cooperative Agreements to State and Local Governments" common rule, or other relevant rules or regulations, for infrastructure assets; provided that the transfer price shall be distributed, as paid, in the following manner: (i) State and local governments shall first recoup in full the unadjusted dollar amount of their portion of total project costs (including any transaction and fix-up costs they incur) associated with the infrastructure asset involved; (ii) if proceeds remain, then the Federal Government shall recoup in full the amount of Federal grant awards associated with the infrastructure asset, less the applicable share of accumulated depreciation on such asset (calculated using the Internal Revenue Service accelerated depreciation schedule for the categories of assets in question); and (iii) finally, the State and local governments shall keep any remaining proceeds.

Sec. 4. Criteria. To the extent permitted by law, the head of an executive department or agency shall approve a request in accordance with section 3(c) of this order only if the grantee: (a) Agrees to use the proceeds described in section 3(c)(iii) of this order only for investment in additional infrastructure assets (after public notice of the proposed investment), or for debt or tax reduction; and

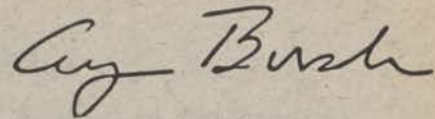
(b) Demonstrates that a market mechanism, legally enforceable agreement, or regulatory mechanism will ensure that: (i) the infrastructure asset or assets will continue to be used for their originally authorized purposes, as long as needed for those purposes, even if the purchaser becomes insolvent or is otherwise hindered from fulfilling the originally authorized purposes; and (ii) user charges will be consistent with any current Federal conditions that protect users and the public by limiting the charges.

Sec. 5. Government-wide Coordination and Review. In implementing Executive Order Nos. 12291 and 12498 and OMB Circular No. A-19, the Office of Management and Budget, to the extent permitted by law and consistent with the provisions of those authorities, shall take action to ensure that the policies of the executive departments and agencies are consistent with the principles, criteria, and requirements of this order. The Office of Management and Budget shall review the results of implementing this order and report thereon to the President 1 year after the date of this order.

Sec. 6. Preservation of Existing Authority. Nothing in this order is in any way intended to limit any existing authority of the heads of executive departments and agencies to approve privatization proposals that are otherwise consistent with law.

Sec. 7. Judicial Review. This order is intended only to improve the internal management of the executive branch, and is not intended to create any right or benefit, substantive or procedural, enforceable by a party against the

United States, its agencies or instrumentalities, its officers or employees, or any other person.

A handwritten signature in dark ink, appearing to read "George H. W. Bush". The signature is written in a cursive, flowing style with a large initial "G".

THE WHITE HOUSE,
April 30, 1992.

[FR Doc. 92-10495

Filed 4-30-92; 4:17 pm]

Billing code 3195-01-M

Presidential Documents

Proclamation 6425 of April 29, 1992

To Amend the Generalized System of Preferences

By the President of the United States of America

A Proclamation

1. Section 504(a)(1) of the Trade Act of 1974, as amended (the 1974 Act) (19 U.S.C. 2464(a)(1)), provides that the President may withdraw, suspend, or limit the application of the duty-free treatment afforded under the Generalized System of Preferences (GSP) with respect to any article or any country after considering the factors set forth in sections 501 and 502(c) of the 1974 Act (19 U.S.C. 2461 and 2462(c)). Pursuant to section 504(a)(1) of the 1974 Act and having considered the factors set forth in sections 501 and 502(c), including, in particular, section 502(c)(5) on the adequate and effective protection of intellectual property rights, I have determined that it is appropriate to suspend the duty-free treatment afforded under the GSP to certain eligible articles that are imported from India, as provided for in the Annex to this proclamation.

2. Section 504(c) of the 1974 Act (19 U.S.C. 2464(c)), provides that beneficiary developing countries are subject to limitations on the preferential treatment afforded under the GSP. Pursuant to 504(c)(1)(B), I have determined that India should no longer receive preferential tariff treatment under the GSP with respect to certain eligible articles, as provided for in the Annex to this proclamation.

3. Section 604 of the 1974 Act (19 U.S.C. 2483) authorizes the President to embody in the Harmonized Tariff Schedule of the United States (HTS) the substance of the relevant provisions of that Act, and of other acts affecting import treatment, and actions thereunder.

NOW, THEREFORE, I, GEORGE BUSH, President of the United States of America, acting under the authority vested in me by the Constitution and the laws of the United States of America, including but not limited to sections 501, 502(c), 504, and 604 of the 1974 Act, do proclaim that:

(1) In order to provide that India should no longer be treated as a beneficiary developing country with respect to certain eligible articles for purposes of the GSP program, the HTS is modified as provided in the Annex to this proclamation.

(2) Any provisions of previous proclamations and Executive orders inconsistent with the provisions of this proclamation are hereby superseded to the extent of such inconsistency.

(3) The amendments made by this proclamation shall be effective with respect to articles both: (i) imported on or after January 1, 1976, and (ii) entered, or withdrawn from warehouse for consumption, on or after 15 days after the date of publication of this proclamation in the Federal Register.

IN WITNESS WHEREOF, I have hereunto set my hand this twenty-ninth day of April, in the year of our Lord nineteen hundred and ninety-two, and of the Independence of the United States of America the two hundred and sixteenth.

George Bush

Billing code 3195-01-M

Annex

Modifications in the Harmonized Tariff Schedule of the United States (HTS)
of an Article's Duty-Free Tariff Treatment with Respect to India
under the Generalized System of Preferences (GSP)

Effective with respect to articles both: (i) imported on or after January 1, 1976, and (ii) entered, or withdrawn from warehouse for consumption, on or after 15 days after the date of publication of this proclamation in the Federal Register:

(a) For the following HTS provisions, in the Rates of Duty 1-Special subcolumn, delete the symbol "A" and insert an "A*" in lieu thereof:

0713.90.10	2827.34.00	2835.31.00	2901.10.30	2906.29.10
2403.91.20	2827.35.00	2835.39.10	2902.50.00	2906.29.20
2801.30.10	2827.36.00	2835.39.50	2903.11.00	2907.11.00
2804.10.00	2827.37.00	2836.10.00	2903.12.00	2907.12.00
2804.21.00	2827.38.00	2836.20.00	2903.13.00	2907.15.10
2804.29.00	2827.39.10	2836.40.10	2903.14.00	2907.19.40
2804.30.00	2827.39.20	2836.40.20	2903.15.00	2907.22.10
2804.40.00	2827.39.30	2836.60.00	2903.16.00	2907.29.10
2805.22.10	2827.39.50	2836.70.00	2903.19.10	2907.29.20
2805.40.00	2827.41.00	2836.91.00	2903.19.50	2908.10.15
2806.20.00	2827.49.10	2836.92.00	2903.21.00	2908.10.20
2810.00.00	2827.49.50	2836.93.00	2903.22.00	2908.90.04
2811.19.10	2827.51.10	2836.99.10	2903.23.00	2908.90.30
2811.19.50	2827.51.20	2836.99.50	2903.29.00	2909.11.00
2811.21.00	2827.59.30	2837.20.10	2903.30.20	2909.19.10
2811.22.10	2827.59.50	2837.20.50	2903.51.00	2909.19.50
2811.23.00	2827.60.20	2838.00.00	2903.59.10	2909.20.00
2811.29.50	2827.60.50	2839.11.00	2903.59.30	2909.30.10
2812.10.50	2828.10.00	2839.19.00	2903.59.50	2909.30.20
2812.90.00	2828.90.00	2839.20.00	2903.61.10	2909.30.30
2813.10.00	2829.19.00	2839.90.00	2903.61.30	2909.41.00
2813.90.50	2829.90.10	2840.11.00	2903.69.05	2909.42.00
2815.30.00	2829.90.50	2840.19.00	2903.69.30	2909.43.00
2816.10.00	2830.10.00	2840.20.00	2904.20.30	2909.44.00
2816.20.00	2830.20.00	2840.30.00	2904.20.50	2909.49.05
2816.30.00	2830.30.00	2841.10.00	2904.90.04	2909.49.20
2818.10.20	2830.90.00	2841.20.00	2904.90.15	2909.49.50
2819.10.00	2831.10.00	2841.30.00	2904.90.50	2909.50.20
2819.90.00	2831.90.00	2841.40.00	2905.11.20	2909.50.40
2820.10.00	2832.10.00	2841.50.00	2905.12.00	2909.60.50
2820.90.00	2832.20.00	2841.60.00	2905.13.00	2910.10.00
2821.10.00	2832.30.10	2841.70.10	2905.14.00	2910.20.00
2821.20.00	2832.30.50	2841.70.50	2905.15.00	2910.30.00
2822.00.00	2833.11.50	2841.90.10	2905.16.00	2910.90.10
2823.00.00	2833.21.00	2841.90.20	2905.19.00	2910.90.50
2824.10.00	2833.23.00	2841.90.30	2905.21.00	2911.00.00
2824.20.00	2833.24.00	2841.90.50	2905.22.10	2912.11.00
2824.90.10	2833.25.00	2842.90.00	2905.22.20	2912.12.00
2824.90.50	2833.26.00	2843.21.00	2905.22.50	2912.13.00
2825.10.00	2833.27.00	2843.29.00	2905.29.00	2912.19.10
2825.20.00	2833.29.10	2843.30.00	2905.31.00	2912.19.20
2825.30.00	2833.29.30	2843.90.00	2905.32.00	2912.19.30
2825.50.10	2833.29.50	2844.10.10	2905.39.10	2912.19.40
2825.50.20	2833.30.00	2844.30.10	2905.39.20	2912.19.50
2825.50.30	2833.40.10	2844.30.50	2905.39.50	2912.29.10
2825.60.00	2833.40.20	2846.10.00	2905.41.00	2912.29.50
2825.70.00	2833.40.50	2846.90.50	2905.42.00	2912.30.20
2825.90.10	2834.10.10	2847.00.00	2905.43.00	2912.30.50
2825.90.20	2834.10.50	2848.10.00	2905.44.00	2912.41.00
2825.90.60	2834.22.00	2849.10.00	2905.49.10	2912.42.00
2826.11.10	2834.29.20	2849.20.20	2905.49.20	2912.49.10
2826.11.50	2834.29.50	2849.90.10	2905.49.50	2912.49.20
2826.19.00	2835.10.00	2849.90.20	2905.50.10	2912.49.50
2826.20.00	2835.21.00	2849.90.50	2905.50.50	2912.50.00
2826.90.00	2835.22.00	2850.00.07	2906.13.10	2912.60.00
2827.10.00	2835.23.00	2850.00.20	2906.13.50	2913.00.50
2827.31.00	2835.24.00	2850.00.50	2906.14.00	2914.12.00
2827.33.00	2835.29.50	2851.00.00	2906.19.00	2914.13.00

Annex (con.)
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(a) (con.)

2914.19.00	2917.19.30	2922.49.40	2933.29.20	2938.90.00
2914.21.20	2917.19.50	2922.49.50	2933.29.45	2939.10.50
2914.22.10	2917.31.00	2922.50.19	2933.29.50	2939.30.00
2914.22.20	2917.32.00	2922.50.50	2933.39.21	2939.50.00
2914.23.00	2917.33.00	2923.10.00	2933.39.23	2939.60.00
2914.29.10	2917.34.00	2923.20.00	2933.39.27	2939.70.00
2914.29.50	2917.35.00	2923.90.00	2933.40.30	2939.90.10
2914.30.00	2917.37.00	2924.10.10	2933.51.10	2939.90.50
2914.41.00	2917.39.20	2924.21.10	2933.59.10	2940.00.00
2914.49.50	2918.11.10	2924.21.15	2933.59.15	2941.10.20
2914.50.50	2918.11.50	2924.21.50	2933.59.18	2941.20.00
2914.69.10	2918.12.00	2924.29.02	2933.59.20	2941.30.00
2914.70.10	2918.13.10	2924.29.04	2933.59.23	2941.50.00
2914.70.50	2918.13.20	2924.29.07	2933.59.30	2941.90.10
2915.11.00	2918.13.30	2924.29.13	2933.59.50	2941.90.50
2915.12.00	2918.13.50	2924.29.14	2933.61.00	2942.00.50
2915.13.10	2918.14.00	2924.29.15	2933.69.00	3001.10.00
2915.13.50	2918.15.10	2924.29.19	2933.71.00	3001.20.00
2915.21.00	2918.15.50	2924.29.25	2933.79.20	3002.90.10
2915.22.00	2918.16.10	2924.29.35	2933.79.30	3003.31.00
2915.23.00	2918.16.50	2924.29.39	2933.79.50	3003.39.10
2915.24.00	2918.17.10	2924.29.42	2933.90.15	3003.40.00
2915.29.00	2918.19.60	2924.29.50	2933.90.18	3003.90.00
2915.32.00	2918.21.10	2925.11.00	2933.90.20	3004.10.10
2915.33.00	2918.22.50	2925.19.50	2933.90.25	3004.20.00
2915.34.00	2918.23.10	2925.20.50	2933.90.31	3004.31.00
2915.35.00	2918.23.20	2926.10.00	2933.90.40	3004.32.00
2915.39.10	2918.29.22	2926.90.21	2933.90.48	3004.39.00
2915.39.20	2918.29.30	2926.90.23	2933.90.50	3004.40.00
2915.39.40	2918.30.50	2926.90.25	2934.10.50	3004.50.30
2915.39.45	2918.90.10	2926.90.27	2934.20.05	3004.50.50
2915.39.47	2918.90.20	2927.00.15	2934.20.10	3004.90.30
2915.39.50	2918.90.35	2927.00.20	2934.20.15	3004.90.60
2915.40.10	2918.90.50	2927.00.30	2934.20.35	3005.10.10
2915.40.50	2919.00.10	2928.00.10	2934.90.10	3005.10.50
2915.50.10	2919.00.50	2928.00.30	2934.90.12	3005.90.10
2915.50.20	2920.10.10	2928.00.50	2934.90.14	3005.90.50
2915.50.50	2920.10.20	2929.10.15	2934.90.16	3006.10.00
2915.60.10	2920.10.50	2929.10.30	2934.90.18	3006.40.00
2915.60.50	2920.90.10	2930.10.00	2934.90.20	3006.50.00
2915.70.00	2920.90.50	2930.20.10	2934.90.25	3006.60.00
2915.90.10	2921.11.00	2930.20.50	2934.90.47	3201.90.10
2915.90.20	2921.12.00	2930.30.00	2934.90.50	3201.90.50
2915.90.50	2921.19.10	2930.40.00	2935.00.05	3202.10.10
2916.12.10	2921.19.50	2930.90.10	2935.00.20	3202.90.50
2916.12.50	2921.21.00	2930.90.30	2935.00.30	3204.19.35
2916.14.00	2921.22.05	2930.90.40	2935.00.31	3204.90.00
2916.15.50	2921.22.50	2930.90.50	2935.00.33	3205.00.20
2916.19.10	2921.29.00	2931.00.25	2935.00.37	3206.10.00
2916.19.20	2921.30.50	2931.00.50	2935.00.43	3206.20.00
2916.19.50	2921.42.23	2932.11.00	2935.00.44	3206.30.00
2916.20.00	2921.42.24	2932.13.00	2936.10.00	3206.41.00
2916.31.10	2921.42.25	2932.19.50	2936.21.00	3206.42.00
2916.31.20	2921.43.18	2932.21.00	2936.22.00	3206.43.00
2916.33.20	2921.49.20	2932.29.10	2936.24.00	3206.49.10
2916.39.08	2921.49.30	2932.29.50	2936.25.00	3206.49.30
2916.39.12	2921.51.20	2932.90.10	2936.27.00	3206.49.50
2916.39.16	2922.11.00	2932.90.20	2936.28.00	3207.10.00
2916.39.20	2922.12.00	2932.90.37	2936.29.15	3207.20.00
2917.11.00	2922.13.00	2932.90.50	2936.29.50	3207.30.00
2917.12.20	2922.19.50	2933.11.00	2936.90.00	3208.10.00
2917.13.00	2922.29.23	2933.19.25	2937.10.00	3208.20.00
2917.14.10	2922.29.25	2933.19.30	2937.21.00	3208.90.00
2917.14.50	2922.29.29	2933.19.35	2937.22.00	3209.10.00
2917.19.15	2922.30.50	2933.19.45	2937.29.00	3209.90.00
2917.19.17	2922.41.00	2933.19.50	2937.91.00	3210.00.00
2917.19.23	2922.42.50	2933.21.00	2938.10.00	3212.10.00

Annex (con.)
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(a) (con.)

3212.90.00	3307.49.00	3504.00.50	3702.95.00	3809.91.00
3213.10.00	3307.90.00	3505.10.00	3703.10.30	3811.11.10
3213.90.00	3401.11.10	3505.20.00	3703.10.60	3811.11.50
3214.10.00	3401.11.50	3506.10.50	3703.20.30	3812.20.10
3215.11.00	3401.19.00	3506.91.00	3703.20.60	3812.30.20
3215.19.00	3401.20.00	3506.99.00	3703.90.30	3813.00.50
3215.90.10	3402.11.10	3507.90.00	3703.90.60	3814.00.20
3215.90.50	3402.11.50	3601.00.00	3706.10.30	3815.90.10
3301.19.10	3402.12.10	3603.00.30	3707.10.00	3815.90.20
3301.24.00	3402.12.50	3603.00.60	3707.90.30	3816.00.00
3301.29.10	3402.13.10	3603.00.90	3707.90.60	3817.10.50
3301.29.20	3402.13.20	3604.10.00	3801.10.10	3823.20.00
3301.30.10	3402.13.50	3604.90.00	3801.30.00	3823.30.00
3302.10.10	3402.19.10	3606.90.60	3801.90.00	3823.60.00
3302.10.20	3402.19.50	3701.10.00	3802.10.00	3823.90.19
3302.90.10	3402.20.10	3701.20.00	3802.90.10	3823.90.22
3302.90.20	3402.90.30	3701.30.00	3802.90.20	3823.90.25
3303.00.20	3402.90.50	3701.91.00	3802.90.50	3823.90.31
3303.00.30	3403.11.40	3701.99.30	3805.10.00	3823.90.32
3304.10.00	3403.11.50	3701.99.60	3806.10.00	3823.90.33
3304.20.00	3403.19.50	3702.10.00	3806.20.00	3823.90.34
3304.30.00	3403.91.10	3702.20.00	3806.30.00	3823.90.36
3304.91.00	3404.20.00	3702.31.00	3807.00.00	3823.90.46
3304.99.00	3405.10.00	3702.32.00	3808.10.10	4104.29.30
3305.10.00	3405.20.00	3702.39.00	3808.10.20	5208.31.20
3305.20.00	3405.30.00	3702.41.00	3808.10.30	5208.32.10
3305.30.00	3405.40.00	3702.42.00	3808.20.10	5208.41.20
3305.90.00	3405.90.00	3702.43.00	3808.20.20	5208.42.10
3306.10.00	3406.00.00	3702.44.00	3808.20.30	5208.51.20
3306.90.00	3407.00.20	3702.51.00	3808.30.10	5208.52.10
3307.10.10	3501.10.10	3702.52.00	3808.30.20	5209.31.30
3307.10.20	3501.90.20	3702.53.00	3808.40.10	5209.41.30
3307.20.00	3501.90.50	3702.54.00	3808.40.50	5310.90.00
3307.30.10	3503.00.10	3702.91.00	3808.90.10	5702.20.10
3307.30.50	3503.00.55	3702.92.00	3808.90.20	6304.99.25
3307.41.00	3504.00.10	3702.93.00	3809.10.00	7012.00.00

(b) General note 3(c)(ii)(D) to the HTS is modified--

(1) by adding, in numerical sequence, the following HTS provisions and the country set opposite them:

0713.90.10	India	2816.30.00	India	2826.11.10	India
2403.91.20	India	2818.10.20	India	2826.11.50	India
2801.30.10	India	2819.10.00	India	2826.19.00	India
2804.10.00	India	2819.90.00	India	2826.20.00	India
2804.21.00	India	2820.10.00	India	2826.90.00	India
2804.29.00	India	2820.90.00	India	2827.10.00	India
2804.30.00	India	2821.10.00	India	2827.31.00	India
2804.40.00	India	2821.20.00	India	2827.33.00	India
2805.22.10	India	2822.00.00	India	2827.34.00	India
2805.40.00	India	2823.00.00	India	2827.35.00	India
2806.20.00	India	2824.10.00	India	2827.36.00	India
2810.00.00	India	2824.20.00	India	2827.37.00	India
2811.19.10	India	2824.90.10	India	2827.38.00	India
2811.19.50	India	2824.90.50	India	2827.39.10	India
2811.21.00	India	2825.10.00	India	2827.39.20	India
2811.22.10	India	2825.20.00	India	2827.39.30	India
2811.23.00	India	2825.30.00	India	2827.39.50	India
2811.29.50	India	2825.50.10	India	2827.41.00	India
2812.10.50	India	2825.50.20	India	2827.49.10	India
2812.90.00	India	2825.50.30	India	2827.49.50	India
2813.10.00	India	2825.60.00	India	2827.51.10	India
2813.90.50	India	2825.70.00	India	2827.51.20	India
2815.30.00	India	2825.90.10	India	2827.59.30	India
2816.10.00	India	2825.90.20	India	2827.59.50	India
2816.20.00	India	2825.90.60	India	2827.60.20	India

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(b)(1) (con.)

2827.60.50	India	2841.20.00	India	2905.21.00	India
2828.10.00	India	2841.30.00	India	2905.22.10	India
2828.90.00	India	2841.40.00	India	2905.22.20	India
2829.19.00	India	2841.50.00	India	2905.22.50	India
2829.90.10	India	2841.60.00	India	2905.29.00	India
2829.90.50	India	2841.70.10	India	2905.31.00	India
2830.10.00	India	2841.70.50	India	2905.32.00	India
2830.20.00	India	2841.90.10	India	2905.39.10	India
2830.30.00	India	2841.90.20	India	2905.39.20	India
2830.90.00	India	2841.90.30	India	2905.39.50	India
2831.10.00	India	2841.90.50	India	2905.41.00	India
2831.90.00	India	2842.90.00	India	2905.42.00	India
2832.10.00	India	2843.21.00	India	2905.43.00	India
2832.20.00	India	2843.29.00	India	2905.44.00	India
2832.30.10	India	2843.30.00	India	2905.49.10	India
2832.30.50	India	2843.90.00	India	2905.49.20	India
2833.11.50	India	2844.10.10	India	2905.49.50	India
2833.21.00	India	2844.30.10	India	2905.50.10	India
2833.23.00	India	2844.30.50	India	2905.50.50	India
2833.24.00	India	2846.10.00	India	2906.13.10	India
2833.25.00	India	2846.90.50	India	2906.13.50	India
2833.26.00	India	2847.00.00	India	2906.14.00	India
2833.27.00	India	2848.10.00	India	2906.19.00	India
2833.29.10	India	2849.10.00	India	2906.29.10	India
2833.29.30	India	2849.20.20	India	2906.29.20	India
2833.29.50	India	2849.90.10	India	2907.11.00	India
2833.30.00	India	2849.90.20	India	2907.12.00	India
2833.40.10	India	2849.90.50	India	2907.15.10	India
2833.40.20	India	2850.00.07	India	2907.19.40	India
2833.40.50	India	2850.00.20	India	2907.22.10	India
2834.10.10	India	2850.00.50	India	2907.29.10	India
2834.10.50	India	2851.00.00	India	2907.29.20	India
2834.22.00	India	2901.10.30	India	2908.10.15	India
2834.29.20	India	2902.50.00	India	2908.10.20	India
2834.29.50	India	2903.11.00	India	2908.90.04	India
2835.10.00	India	2903.12.00	India	2908.90.30	India
2835.21.00	India	2903.13.00	India	2909.11.00	India
2835.22.00	India	2903.14.00	India	2909.19.10	India
2835.23.00	India	2903.15.00	India	2909.19.50	India
2835.24.00	India	2903.16.00	India	2909.20.00	India
2835.29.50	India	2903.19.10	India	2909.30.10	India
2835.31.00	India	2903.19.50	India	2909.30.20	India
2835.39.10	India	2903.21.00	India	2909.30.30	India
2835.39.50	India	2903.22.00	India	2909.41.00	India
2836.10.00	India	2903.23.00	India	2909.42.00	India
2836.20.00	India	2903.29.00	India	2909.43.00	India
2836.40.10	India	2903.30.20	India	2909.44.00	India
2836.40.20	India	2903.51.00	India	2909.49.05	India
2836.60.00	India	2903.59.10	India	2909.49.20	India
2836.70.00	India	2903.59.30	India	2909.49.50	India
2836.91.00	India	2903.59.50	India	2909.50.20	India
2836.92.00	India	2903.61.10	India	2909.50.40	India
2836.93.00	India	2903.61.30	India	2909.60.50	India
2836.99.10	India	2903.69.05	India	2910.10.00	India
2836.99.50	India	2903.69.30	India	2910.20.00	India
2837.20.10	India	2904.20.30	India	2910.30.00	India
2837.20.50	India	2904.20.50	India	2910.90.10	India
2838.00.00	India	2904.90.04	India	2910.90.50	India
2839.11.00	India	2904.90.15	India	2911.00.00	India
2839.19.00	India	2904.90.50	India	2912.11.00	India
2839.20.00	India	2905.11.20	India	2912.12.00	India
2839.90.00	India	2905.12.00	India	2912.13.00	India
2840.11.00	India	2905.13.00	India	2912.19.10	India
2840.19.00	India	2905.14.00	India	2912.19.20	India
2840.20.00	India	2905.15.00	India	2912.19.30	India
2840.30.00	India	2905.16.00	India	2912.19.40	India
2841.10.00	India	2905.19.00	India	2912.19.50	India

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(b)(1) (con.)

2912.29.10	India	2916.31.20	India	2921.43.18	India
2912.29.50	India	2916.33.20	India	2921.49.20	India
2912.30.20	India	2916.39.08	India	2921.49.30	India
2912.30.50	India	2916.39.12	India	2921.51.20	India
2912.41.00	India	2916.39.16	India	2922.11.00	India
2912.42.00	India	2916.39.20	India	2922.12.00	India
2912.49.10	India	2917.11.00	India	2922.13.00	India
2912.49.20	India	2917.12.20	India	2922.19.50	India
2912.49.50	India	2917.13.00	India	2922.29.23	India
2912.50.00	India	2917.14.10	India	2922.29.25	India
2912.60.00	India	2917.14.50	India	2922.29.29	India
2913.00.50	India	2917.19.15	India	2922.30.50	India
2914.12.00	India	2917.19.17	India	2922.41.00	India
2914.13.00	India	2917.19.23	India	2922.42.50	India
2914.19.00	India	2917.19.30	India	2922.49.40	India
2914.21.20	India	2917.19.50	India	2922.49.50	India
2914.22.10	India	2917.31.00	India	2922.50.19	India
2914.22.20	India	2917.32.00	India	2922.50.50	India
2914.23.00	India	2917.33.00	India	2923.10.00	India
2914.29.10	India	2917.34.00	India	2923.20.00	India
2914.29.50	India	2917.35.00	India	2923.90.00	India
2914.30.00	India	2917.37.00	India	2924.10.10	India
2914.41.00	India	2917.39.20	India	2924.21.10	India
2914.49.50	India	2918.11.10	India	2924.21.15	India
2914.50.50	India	2918.11.50	India	2924.21.50	India
2914.69.10	India	2918.12.00	India	2924.29.02	India
2914.70.10	India	2918.13.10	India	2924.29.04	India
2914.70.50	India	2918.13.20	India	2924.29.07	India
2915.11.00	India	2918.13.30	India	2924.29.13	India
2915.12.00	India	2918.13.50	India	2924.29.14	India
2915.13.10	India	2918.14.00	India	2924.29.15	India
2915.13.50	India	2918.15.10	India	2924.29.19	India
2915.21.00	India	2918.15.50	India	2924.29.25	India
2915.22.00	India	2918.16.10	India	2924.29.35	India
2915.23.00	India	2918.16.50	India	2924.29.39	India
2915.24.00	India	2918.17.10	India	2924.29.42	India
2915.29.00	India	2918.19.60	India	2924.29.50	India
2915.32.00	India	2918.21.10	India	2925.11.00	India
2915.33.00	India	2918.22.50	India	2925.19.50	India
2915.34.00	India	2918.23.10	India	2925.20.50	India
2915.35.00	India	2918.23.20	India	2926.10.00	India
2915.39.10	India	2918.29.22	India	2926.90.21	India
2915.39.20	India	2918.29.30	India	2926.90.23	India
2915.39.40	India	2918.30.50	India	2926.90.25	India
2915.39.45	India	2918.90.10	India	2926.90.27	India
2915.39.47	India	2918.90.20	India	2927.00.15	India
2915.39.50	India	2918.90.35	India	2927.00.20	India
2915.40.10	India	2918.90.50	India	2927.00.30	India
2915.40.50	India	2919.00.10	India	2928.00.10	India
2915.50.10	India	2919.00.50	India	2928.00.30	India
2915.50.20	India	2920.10.10	India	2928.00.50	India
2915.50.50	India	2920.10.20	India	2929.10.15	India
2915.60.10	India	2920.10.50	India	2929.10.30	India
2915.60.50	India	2920.90.10	India	2930.10.00	India
2915.70.00	India	2920.90.50	India	2930.20.10	India
2915.90.10	India	2921.11.00	India	2930.20.50	India
2915.90.20	India	2921.12.00	India	2930.30.00	India
2915.90.50	India	2921.19.10	India	2930.40.00	India
2916.12.10	India	2921.19.50	India	2930.90.10	India
2916.12.50	India	2921.21.00	India	2930.90.30	India
2916.14.00	India	2921.22.05	India	2930.90.40	India
2916.15.50	India	2921.22.50	India	2930.90.50	India
2916.19.10	India	2921.29.00	India	2931.00.25	India
2916.19.20	India	2921.30.50	India	2931.00.50	India
2916.19.50	India	2921.42.23	India	2932.11.00	India
2916.20.00	India	2921.42.24	India	2932.13.00	India
2916.31.10	India	2921.42.25	India	2932.19.50	India

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(b)(1) (con.)

2932.21.00	India	2936.22.00	India	3206.43.00	India
2932.29.10	India	2936.24.00	India	3206.49.10	India
2932.29.50	India	2936.25.00	India	3206.49.30	India
2932.90.10	India	2936.27.00	India	3206.49.50	India
2932.90.20	India	2936.28.00	India	3207.10.00	India
2932.90.37	India	2936.29.15	India	3207.20.00	India
2932.90.50	India	2936.29.50	India	3207.30.00	India
2933.11.00	India	2936.90.00	India	3208.10.00	India
2933.19.25	India	2937.10.00	India	3208.20.00	India
2933.19.30	India	2937.21.00	India	3208.90.00	India
2933.19.35	India	2937.22.00	India	3209.10.00	India
2933.19.45	India	2937.29.00	India	3209.90.00	India
2933.19.50	India	2937.91.00	India	3210.00.00	India
2933.21.00	India	2938.10.00	India	3212.10.00	India
2933.29.20	India	2938.90.00	India	3212.90.00	India
2933.29.45	India	2939.10.50	India	3213.10.00	India
2933.29.50	India	2939.30.00	India	3213.90.00	India
2933.39.21	India	2939.50.00	India	3214.10.00	India
2933.39.23	India	2939.60.00	India	3215.11.00	India
2933.39.27	India	2939.70.00	India	3215.19.00	India
2933.40.30	India	2939.90.10	India	3215.90.10	India
2933.51.10	India	2939.90.50	India	3215.90.50	India
2933.59.10	India	2940.00.00	India	3301.19.10	India
2933.59.15	India	2941.10.20	India	3301.24.00	India
2933.59.18	India	2941.20.00	India	3301.29.10	India
2933.59.20	India	2941.30.00	India	3301.29.20	India
2933.59.23	India	2941.50.00	India	3301.30.10	India
2933.59.30	India	2941.90.10	India	3302.10.10	India
2933.59.50	India	2941.90.50	India	3302.10.20	India
2933.61.00	India	2942.00.50	India	3302.90.10	India
2933.69.00	India	3001.10.00	India	3302.90.20	India
2933.71.00	India	3001.20.00	India	3303.00.20	India
2933.79.20	India	3002.90.10	India	3303.00.30	India
2933.79.30	India	3003.31.00	India	3304.10.00	India
2933.79.50	India	3003.39.10	India	3304.20.00	India
2933.90.15	India	3003.40.00	India	3304.30.00	India
2933.90.18	India	3003.90.00	India	3304.91.00	India
2933.90.20	India	3004.10.10	India	3304.99.00	India
2933.90.25	India	3004.20.00	India	3305.10.00	India
2933.90.31	India	3004.31.00	India	3305.20.00	India
2933.90.40	India	3004.32.00	India	3305.30.00	India
2933.90.48	India	3004.39.00	India	3305.90.00	India
2933.90.50	India	3004.40.00	India	3306.10.00	India
2934.10.50	India	3004.50.30	India	3306.90.00	India
2934.20.05	India	3004.50.50	India	3307.10.10	India
2934.20.10	India	3004.90.30	India	3307.10.20	India
2934.20.15	India	3004.90.60	India	3307.20.00	India
2934.20.35	India	3005.10.10	India	3307.30.10	India
2934.90.10	India	3005.10.50	India	3307.30.50	India
2934.90.12	India	3005.90.10	India	3307.41.00	India
2934.90.14	India	3005.90.50	India	3307.49.00	India
2934.90.16	India	3006.10.00	India	3307.90.00	India
2934.90.18	India	3006.40.00	India	3401.11.10	India
2934.90.20	India	3006.50.00	India	3401.11.50	India
2934.90.25	India	3006.60.00	India	3401.19.00	India
2934.90.47	India	3201.90.10	India	3401.20.00	India
2934.90.50	India	3201.90.50	India	3402.11.10	India
2935.00.05	India	3202.10.10	India	3402.11.50	India
2935.00.20	India	3202.90.50	India	3402.12.10	India
2935.00.30	India	3204.19.35	India	3402.12.50	India
2935.00.31	India	3204.90.00	India	3402.13.10	India
2935.00.33	India	3205.00.20	India	3402.13.20	India
2935.00.37	India	3206.10.00	India	3402.13.50	India
2935.00.43	India	3206.20.00	India	3402.19.10	India
2935.00.44	India	3206.30.00	India	3402.19.50	India
2936.10.00	India	3206.41.00	India	3402.20.10	India
2936.21.00	India	3206.42.00	India	3402.90.30	India

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(b)(1) (con.)

3402.90.50	India	3702.39.00	India	3808.40.10	India
3403.11.40	India	3702.41.00	India	3808.40.50	India
3403.11.50	India	3702.42.00	India	3808.90.10	India
3403.19.50	India	3702.43.00	India	3808.90.20	India
3403.91.10	India	3702.44.00	India	3809.10.00	India
3404.20.00	India	3702.51.00	India	3809.91.00	India
3405.10.00	India	3702.52.00	India	3811.11.10	India
3405.20.00	India	3702.53.00	India	3811.11.50	India
3405.30.00	India	3702.54.00	India	3812.20.10	India
3405.40.00	India	3702.91.00	India	3812.30.20	India
3405.90.00	India	3702.92.00	India	3813.00.50	India
3406.00.00	India	3702.93.00	India	3814.00.20	India
3407.00.20	India	3702.95.00	India	3815.90.10	India
3501.10.10	India	3703.10.30	India	3815.90.20	India
3501.90.20	India	3703.10.60	India	3816.00.00	India
3501.90.50	India	3703.20.30	India	3817.10.50	India
3503.00.10	India	3703.20.60	India	3823.20.00	India
3503.00.55	India	3703.90.30	India	3823.30.00	India
3504.00.10	India	3703.90.60	India	3823.60.00	India
3504.00.50	India	3706.10.30	India	3823.90.19	India
3505.10.00	India	3707.10.00	India	3823.90.22	India
3505.20.00	India	3707.90.30	India	3823.90.25	India
3506.10.50	India	3707.90.60	India	3823.90.31	India
3506.91.00	India	3801.10.10	India	3823.90.32	India
3506.99.00	India	3801.30.00	India	3823.90.33	India
3507.90.00	India	3801.90.00	India	3823.90.34	India
3601.00.00	India	3802.10.00	India	3823.90.36	India
3603.00.30	India	3802.90.10	India	3823.90.46	India
3603.00.60	India	3802.90.20	India	4104.29.30	India
3603.00.90	India	3802.90.50	India	5208.31.20	India
3604.10.00	India	3805.10.00	India	5208.32.10	India
3604.90.00	India	3806.10.00	India	5208.41.20	India
3606.90.60	India	3806.20.00	India	5208.42.10	India
3701.10.00	India	3806.30.00	India	5208.51.20	India
3701.20.00	India	3807.00.00	India	5208.52.10	India
3701.30.00	India	3808.10.10	India	5209.31.30	India
3701.91.00	India	3808.10.20	India	5209.41.30	India
3701.99.30	India	3808.10.30	India	5310.90.00	India
3701.99.60	India	3808.20.10	India	5702.20.10	India
3702.10.00	India	3808.20.20	India	6304.99.25	India
3702.20.00	India	3808.20.30	India	7012.00.00	India
3702.31.00	India	3808.30.10	India		
3702.32.00	India	3808.30.20	India		

(2) by adding, in alphabetical order, the country opposite the following HTS subheadings:

2804.69.10	India	2918.22.10	India	3203.00.50	India
2825.90.15	India	2918.90.30	India	3207.40.10	India
2827.59.05	India	2929.90.50	India	3301.12.00	India
2903.40.00	India	2933.39.25	India	3407.90.10	India
2903.59.40	India	2933.40.10	India	3823.90.40	India
2906.11.00	India	2933.90.47	India		
2915.31.00	India	2937.92.10	India		

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Filed 4-30-92; 2:11 pm]

Billing code 3195-01-C

Presidential Documents

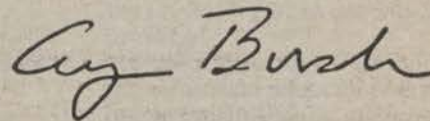
Presidential Determination No. 92-12 of January 31, 1992

Renewal of Trade Agreement With the People's Republic of China

Memorandum for the United States Trade Representative

Pursuant to my authority under subsection 405(b)(1) of the Trade Act of 1974 (19 U.S.C. 2435(b)(1)), I have determined that actual or foreseeable reductions in United States tariffs and nontariff barriers to trade resulting from multilateral negotiations are satisfactorily reciprocated by the People's Republic of China. I have further found that a satisfactory balance of concessions in trade and services has been maintained during the life of the Agreement on Trade Relations between the United States of America and the People's Republic of China.

You are authorized and directed to publish this memorandum in the **Federal Register**.



THE WHITE HOUSE,
Washington, January 31, 1992.

[FR Doc. 92-10463

Filed 4-30-92; 2:32 pm]

Billing code 3195-01-M

Rules and Regulations

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Monday, May 4, 1992

This section of the FEDERAL REGISTER contains regulatory documents having general applicability and legal effect, most of which are keyed to and codified in the Code of Federal Regulations, which is published under 50 titles pursuant to 44 U.S.C. 1510.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 91-ANE-43; Amendment 39-8086, AD 92-24-01]

Airworthiness Directives; General Electric Company CF6-6 Turbofan Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD) applicable to General Electric (GE) CF6-6 series turbofan engines that currently requires initial and repetitive inspection of the compressor rear frame (CRF) manifold port plug weld for cracks, and removal from service of certain CRF casings. This amendment increases the number of CRF casings to be inspected, reduces inspection intervals, and requires CRF rework. This amendment is prompted by the results of further investigation of a CRF outer case failure. The actions specified by this AD are intended to prevent rupture of the CRF casing that could result in an engine shutdown, aircraft damage, and an aborted takeoff.

DATES: Effective May 26, 1992.

The incorporation by reference of General Electric CF6-6 Service Bulletin 72-971, dated October 2, 1990, was approved by the Director of the Federal Register on December 11, 1990.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 26, 1992.

Comments for inclusion in the Rules Docket must be received on or before July 6, 1992.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Assistant Chief Counsel, Attention: Rules Docket No. 91-ANE-43, New England Region, 12 New England Executive Park, Burlington, Massachusetts 01803-5299.

The service information referenced in this AD may be obtained from General Electric Aircraft Engines, Technical Publications Department, 1 Neumann Way, Cincinnati, Ohio 45215. This information may be examined at the FAA, New England Region, Office of the Assistant Chief Counsel, 12 New England Executive Park, Burlington, Massachusetts; or at the Office of the Federal Register, 1100 L Street NW., room 8401, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Robert Ganley, Engine Certification Office, ANE-140, Engine and Propeller Directorate, Aircraft Certification Service, FAA, New England Region, 12 New England Executive Park, Burlington, Massachusetts 01803-5299; telephone (617) 272-5047; fax (617) 270-2412.

SUPPLEMENTARY INFORMATION: On November 21, 1990, Airworthiness Directive (AD) 90-24-12, Amendment 39-8086 (55 FR 48591) was published in the Federal Register. This AD requires initial and repetitive inspections of the General Electric (GE) CF6-6 series turbofan engine compressor rear frame (CRF), and removal of cracked CRF's in accordance with GE CF6-6 Service Bulletin (SB) 72-971, dated October 2, 1990. That action was prompted by the report of a CRF outer case rupture at the compressor discharge pressure (CDP) manifold port plug. That failure resulted in an engine shutdown, aborted takeoff and damage to the aircraft, specifically the engine pylon and horizontal stabilizer.

A subsequent metallurgical examination of the fracture surface revealed lack of fusion in the manifold port plug electron beam weld. A low cycle fatigue crack initiated and propagated to critical length resulting in an axial rupture of the CRF outer case at the twelve o'clock position. That condition, if not corrected, can result in an engine shutdown, aircraft damage, and an aborted takeoff.

Since issuance of AD 90-24-12, the FAA has determined that additional CRF's are affected, and that these

additional CRF's must be inspected. This AD will require compliance with GE CF6-6 SB 72-971, Revision 2, dated August 27, 1991, which lists the additional affected part numbers. The revised service bulletin also reduces certain inspection intervals, but provides that the intervals may be increased if an optional ultrasonic inspection is performed in addition to the required inspection methods. Finally, as a terminating action to this AD, a shop level inspection and rework of the CRF as described in GE CF6-6 SB 72-977, dated March 15, 1991, have been added that eliminates the repetitive installed inspection requirement.

Since an unsafe condition has been identified that is likely to exist or develop on other engines of this same type design, this AD supersedes AD 90-24-12 to require initial and repetitive inspections of the CRF and removal of cracked CRF's. The actions are required to be accomplished in accordance with the service bulletins described previously.

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to submit such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified under the caption "ADDRESSES." All communication received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted

will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 91-ANE-43". The postcard will be date stamped and returned to the commenter.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

The FAA has determined that this regulation is an emergency regulation and that it is not considered to be major under Executive Order 12291. It is impracticable for the agency to follow the procedures of Executive Order 12291 with respect to this rule since the rule must be issued immediately to correct an unsafe condition in aircraft. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption "ADDRESSES."

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration (FAA) amends 14 CFR part 39 of the Federal Aviation Regulations (FAR) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-6808 (55 FR 48591, November 21, 1990), and by adding the following new airworthiness directive (AD):

91-24-01—General Electric Company:

Amendment 39-8086, Docket No. 91-ANE-43. Supersedes AD 90-24-12, Amendment 39-6808.

Applicability: General Electric Company (GE) CF6-6 series turbofan engines installed on but not limited to McDonnell-Douglas DC-10 aircraft.

Compliance: Required as indicated, unless accomplished previously.

To prevent rupture of the compressor rear frame (CRF) casing that could result in an engine shutdown, aircraft damage, and aborted takeoff, accomplish the following:

(a) At the next scheduled open cowl check, but not later than 100 cycles in service (CIS) after the effective date of this AD, inspect the CRF outer case compressor discharge pressure (CDP) manifold port plug weld area for cracks in accordance with GE CF6-6 Service Bulletin (SB) 72-971, Revision 2, dated August 27, 1991.

(b) Remove from service, prior to further flight, CRF casings which exceed the serviceable limits specified in table 1 or table 2, as applicable, of GE CF6-6 SB 72-971, Revision 2, dated August 27, 1991.

(c) For engines with CRF casings inspected after December 11, 1990 (the effective date of AD 90-24-12), reinspect and remove from service CRF casings in accordance with the inspection intervals and crack limits specified in table 1 of GE CF6-6 SB 72-971, dated October 2, 1990, until the first scheduled inspection after the effective date of this AD.

(d) Thereafter, reinspect and remove from service CRF casings in accordance with the inspection intervals and crack limits specified in table 1 or table 2, as applicable, of GE CF6-6 SB 72-971, Revision 2, dated August 27, 1991.

(e) At the next shop visit, but not later than 4,500 CIS after the effective date of this AD, perform a visual, fluorescent-penetrant, and radiographic inspection of the CRF, and rework the CRF in accordance with GE CF6-6 SB 72-977, dated March 15, 1991.

(f) For the purpose of this AD, a shop visit is defined as the CRF exposed at the piece-part level.

(g) Compliance with paragraph (e) of this AD constitutes terminating action for the inspection requirements of paragraphs (a), (b), (c), and (d).

(h) An alternative method of compliance or adjustment of the compliance time, which provides an acceptable level of safety, may be used when approved by the Manager, Engine Certification Office, ANE-140, Engine and Propeller Directorate, Aircraft Certification Service, FAA, New England Region, 12 New England Executive Park, Burlington, Massachusetts. The request shall

be forwarded through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Engine Certification Office.

(i) Special flight permits may be issued in accordance with FAR 21.197 and 21.199 to operate the airplane to a location where the requirements of this AD can be accomplished.

(j) The reinspections and removal of CRF casings shall be done in accordance with General Electric CF6-6 Service Bulletin 72-971, dated October 2, 1990. This incorporation by reference was previously approved by the Director of the Federal Register at 55 FR 48591 (December 11, 1990) in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies of General Electric service bulletins may be obtained from General Electric Aircraft Engines, Technical Publications Department, 1 Neumann Way, Cincinnati, Ohio 45215. Copies may be inspected at the FAA, New England Region, Office of the Assistant Chief Counsel, 12 New England Executive Park, room 311, Burlington, Massachusetts, or at the Office of the Federal Register, 1100 L Street NW., room 8401, Washington, DC.

(k) The inspections and rework shall be done in accordance with the following General Electric Company documents:

Document No.	Page No.	Issue/Rev.	Date
GE CF6-6 SB 72-971.	1-11	Revision 2	8/27/91
Total: 12 pages.	12	Original	10/2/90
GE CF6-6 SB 72-977.	1-34	Original	3/15/91
Total: 34 pages.			

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies of General Electric service bulletins may be obtained from General Electric Aircraft Engines, Technical Publications Department, 1 Neumann Way, Cincinnati, Ohio 45215. Copies may be inspected at the FAA, New England Region, Office of the Assistant Chief Counsel, 12 New England Executive Park, room 311, Burlington, Massachusetts, or at the Office of the Federal Register, 1100 L Street NW., room 8401, Washington, D.C.

(l) This amendment becomes effective on May 26, 1992.

Issued in Burlington, Massachusetts, on March 31, 1992.

Mark C. Fulmer,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 92-10288 Filed 5-1-92; 8:45 am]

BILLING CODE 4910-13-M

14 CFR Part 39

[Docket NO. 91-ANE-36; Amendment 39-8088, AD 91-24-03]

Airworthiness Directives; Pratt & Whitney Canada PW100 Series Turboprop Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD) applicable to Pratt & Whitney Canada (PWC) PW123 and PW124B turboprop engines that currently requires repetitive turbomachinery magnetic chip detector (MCD) inspections to identify impending bearing failures, and a one-time inspection of the rear inlet case oil system. This amendment includes the repetitive MCD inspections of the AD being superseded, increases the model effectivity to include the PW125B and PW126A turboprop engines, and requires the incorporation of an improved design low pressure (LP) rotor balancing assembly as a terminating action to the inspection program. This amendment is prompted by the availability of the improved design LP rotor balancing assembly and the need to increase model effectivity. The actions specified by this AD are intended to prevent a low pressure turbine (LPT) overspeed, uncontained disk failure, and subsequent damage to the aircraft.

DATES: Effective May 26, 1992.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 26, 1992.

Comments for inclusion in the Rules Docket must be received on or before July 6, 1992.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Assistant Chief Counsel, Attention: Rules Docket No. 91-ANE-36, New England Region, 12 New England Executive Park, Burlington, Massachusetts 01803-5299.

The service information referenced in this AD may be obtained from Pratt & Whitney Canada, Technical Publications Department, 1000 Marie Victorin, Longueuil, Quebec J4C 1A1. This information may be examined at the FAA, New England Region, Office of the Assistant Chief Counsel, 12 New England Executive Park, Burlington, Massachusetts; or at the Office of the

Federal Register, 1100 L Street NW., room 8401, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Robert Ganley, Engine Certification Office, ANE-140, Engine and Propeller Directorate, Aircraft Certification Service, FAA, New England Region, 12 New England Executive Park, Burlington, Massachusetts 01803-5299; telephone (617) 272-5047; fax (617) 270-2412.

SUPPLEMENTARY INFORMATION: On November 6, 1990, the Federal Aviation Administration (FAA) issued Airworthiness Directive (AD) 90-24-05, Amendment 39-6812 (55 FR 47848, November 16, 1990), to require repetitive magnetic chip detector (MCD) inspections to identify impending bearing failures, and a one-time inspection of the rear inlet case oil system. That action was prompted by a reported No. 2 bearing failure which caused the low pressure turbine (LPT) integral stub shaft to fracture, resulting in an LPT overspeed and an uncontained LPT disk failure. The bearing failure was caused by oil starvation due to an obstructed oil jet. This condition, if not corrected, can result in an LPT overspeed, uncontained disk failure, and subsequent damage to the aircraft.

One commenter responded to that final rule (AD 90-24-05). The commenter has no objection to the initial inspection for new engines. However, this commenter does not agree with the repetitive inspection intervals. This commenter states that normal maintenance inspection intervals will adequately detect developing problems on the PW123 and PW124B engines, and that additional repetitive inspections are a burden and do very little, if any, good in improving the level of safety. The FAA does not concur with the commenter's assertions. The data received was not sufficient to change the rule. However, the FAA will take into consideration a change to the rule when sufficient data has been submitted to justify a change. Therefore, the reinspection intervals will remain in the rule.

Since issuance of AD 90-24-05, the FAA has determined that an improved design low pressure (LP) rotor balancing assembly is available which would eliminate the need for repetitive inspections of the turbomachinery MCD and the one-time inspection of the rear inlet case oil system as required by the existing AD. This AD also increases the number of engines affected. The PW125B and PW126A model engines have a similar LPT design to the engine which experienced the reported No. 2

bearing failure. The FAA has therefore determined that these additional PW100 models must be included in the AD.

The FAA has reviewed and approved the technical contents of Pratt & Whitney Canada (PWC) Service Bulletin (SB) No. 21018, Revision 2, dated November 25, 1991, that describes the improved design LP rotor balancing assembly and lists the increased number of engines affected; and PWC SB No. 20938, Revision 2, dated November 18, 1991, that describes the repetitive inspections of the turbomachinery MCD.

Since an unsafe condition has been identified that is likely to exist or develop on other engines of the same type design, this AD supersedes AD 90-24-05 to require incorporation of the improved design LP rotor balancing assembly, increase the number of engines affected by including additional engine models, PW125B and PW126A, and continue the previous MCD inspection requirements. Since there is no change in the inspection requirements of this AD from those required by AD 90-24-05, credit will be given for inspections performed in accordance with the requirements of AD 90-24-05. The actions are required to be accomplished in accordance with the service bulletins described previously.

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to submit such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified under the caption "ADDRESSES." All communication received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted

will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 91-ANE-36". The postcard will be date stamped and returned to the commenter.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

The FAA has determined that this regulation is an emergency regulation and that it is not considered to be major under Executive Order 12291. It is impracticable for the agency to follow the procedures of Executive Order 12291 with respect to this rule since the rule must be issued immediately to correct an unsafe condition in aircraft. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption "ADDRESSES."

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration (FAA) amends 14 CFR part 39 of the Federal Aviation Regulations (FAR) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-6812 (55 FR 47848, November 6, 1990), and by adding the following new airworthiness directive (AD):

91-24-03—Pratt & Whitney Canada:

Amendment 39-8088, Docket No. 91-ANE-36. Supersedes AD 90-24-05, Amendment 39-6812.

Applicability: Pratt & Whitney Canada (PWC) PW123, PW124B, PW125B, and PW126A turboprop engines installed on but not limited to DeHavilland Dash 8 Series 300, Aerospatiale ATR72, Fokker 50, and British Aerospace ATP aircraft.

Compliance: Required as indicated, unless accomplished previously.

To prevent a low pressure turbine (LPT) overspeed, that could result in uncontained disk failure, and subsequent damage to the aircraft, accomplish the following:

(a) For PW123 and PW124B engines that have not been modified in accordance with any revision level of PWC Service Bulletin (SB) 21018, prior to the effective date of this AD, accomplish the following:

(1) Within 5 days after the effective date of this AD, perform a turbomachinery magnetic chip detector (MCD) continuity check in accordance with PWC SB 20938, Revision 2, dated November 18, 1991.

(2) Within 5 days after the effective date of this AD, perform a turbomachinery MCD functional check in accordance with PWC SB 20938, Revision 2, dated November 18, 1991.

(3) Thereafter, at intervals not to exceed 25 flight hours since last inspection, perform the turbomachinery MCD continuity check specified in paragraph (a)(1) of this AD.

(4) Thereafter, at intervals not to exceed 300 flight hours since last inspection, perform the turbomachinery MCD functional check specified in paragraph (a)(2) of this AD.

(b) In addition to the requirements of paragraph (a) of this AD, for PW123 engines that have not been modified in accordance with any revision level of PWC SB 21018, prior to the effective date of this AD, accomplish the following:

(1) Within 5 days after the effective date of this AD, perform a turbomachinery MCD airframe circuitry check in accordance with PWC SB 20938, Revision 2, dated November 18, 1991.

(2) Thereafter, at intervals not to exceed 300 flight hours since last inspection, perform turbomachinery MCD airframe circuitry check specified in paragraph (b)(1) of this AD.

(c) For PW125B and PW126A engines have not been modified in accordance with any revision level of PWC SB 21018, prior to the effective date of this AD, accomplish the following:

(1) Within the next 125 flight hours after the effective date of this AD, unless previously accomplished within the last 875 flight hours, perform a turbomachinery MCD operational check in accordance with the applicable engine maintenance manual.

Note: Further information on the turbomachinery MCD operational check specified in paragraph (c)(1) can be found in PWC Maintenance Manual Part Number 3034932 for the PW125B, and PWC Maintenance Manual Part Number 3034922 for the PW126A.

(2) Within the next 125 flight hours after the effective date of this AD, unless previously accomplished within the last 875 flight hours, perform an operational check of the turbomachinery MCD airframe circuitry and indicating system in accordance with the applicable aircraft maintenance manual.

NOTE: Further information on the turbomachinery MCD airframe circuitry and indicating system specified in paragraph (c)(2) of this AD can be found in Fokker 50 Maintenance Manual, Chapter 77-33-00, for the PW125B, and BAe ATP Aircraft Maintenance Manual, Chapter 79-33-00, for the PW126A.

(3) Thereafter, at intervals not to exceed 1,000 flight hours since last inspection, perform the turbomachinery MCD operational check specified in paragraph (c)(1) of this AD.

(4) Thereafter, at intervals not to exceed 1,000 flight hours since last inspection, perform the operational check of the turbomachinery MCD airframe circuitry and indicating system in paragraph (c)(2) of this AD.

(d) The initial inspections of paragraphs (a)(1), (a)(2), and (b)(1) of this AD need not be accomplished for those PW123 and PW124B engines previously inspected in accordance with AD 90-24-05.

(e) For engines that have not been modified in accordance with any revision level of PWC SB 21018, prior to the effective date of this AD, incorporate a new low pressure rotor balancing assembly in accordance with PWC SB 21018, Revision 2, dated November 25, 1991, at the next engine shop visit or by June 30, 1994, whichever occurs first. The incorporation of a new low pressure rotor balancing assembly in accordance with this paragraph constitutes a terminating action for paragraphs (a), (b), or (c), as applicable, of this AD.

(f) For the purpose of this AD, an engine shop visit is defined as the induction of an engine into a maintenance facility for the conduct of any type of maintenance.

(g) An alternative method of compliance or adjustment of the compliance time, which provides an acceptable level of safety, may be used when approved by the Manager, Engine Certification Office, ANE-140, Engine and Propeller Directorate, Aircraft Certification Service, FAA, New England Region, 12 New England Executive Park, Burlington, Massachusetts. The request shall be forwarded through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Engine Certification Office.

(h) Special flight permits may be issued in accordance with FAR 21.197 and 21.199 to operate the airplane to a location where the requirements of this AD can be accomplished.

(i) The inspections and modification shall be done in accordance with the following Pratt & Whitney Canada service documents:

Document number	Page no.	Issue/revision	Date
PWC SB 21018R2	1-36	Revision 2	11/25/91
Total: 36 pages			
PWC SB 20938R2	1-8	Revision 2	11/18/91
Total: 8 pages			

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR Part 51. Copies may be obtained from Pratt & Whitney Canada, Technical Publications Department, 1000 Marie Victorin, Longueuil, Quebec J4G 1A1. Copies may be inspected at the FAA, New England Region, Office of the Assistant Chief Counsel, 12 New England Executive Park, room 311, Burlington, Massachusetts, or at the Office of the Federal Register, 1100 L Street NW., room 8401, Washington, DC.

(j) This amendment becomes effective on May 28, 1992.

Issued in Burlington, Massachusetts, on April 2, 1992.

Jack A. Sain,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 92-10289 Filed 5-1-92; 8:45 am]

BILLING CODE 4910-13-M

14 CFR Part 71

[Airspace Docket No. 92-AEA-4]

Alteration of VOR Federal Airways, Compulsory Reporting Point, and Jet Route J-191; DE

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: These amendments alter the descriptions of Federal Airways V-16, V-29, V-157, V-213, V-268, V-379; Jet Route J-191; and the compulsory reporting point in the State of Delaware by renaming the Kenton, DE, VHF Omnidirectional Range/Tactical Air Navigation (VORTAC) to Smyrna, DE, VORTAC. There will be no physical change to the airway structure as these amendments will only reflect a name change from Kenton, DE, VORTAC to Smyrna, DE, VORTAC. The renaming of the Kenton VORTAC is necessary to eliminate similar sounding names for navigational facilities in the Washington Air Route Traffic Control Center's areas.

EFFECTIVE DATE: 0901 u.t.c., June 25, 1992.

FOR FURTHER INFORMATION CONTACT: Patricia P. Crawford, Airspace and Obstruction Evaluation Branch (ATP-240), Airspace-Rules and Aeronautical Information Division, Air Traffic Rules and Procedures Service, Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone: (202) 267-9255.

The Rule

These amendments to part 71 of the Federal Aviation Regulations alter the descriptions of V-16, V-29, V-157, V-213, V-268, V-379; Jet Route J-191; and the Kenton, DE, Compulsory Reporting Point by renaming the Kenton VORTAC to the Smyrna VORTAC. I find that notice and public procedure under 5 U.S.C. 553(b) are unnecessary because these actions are inconsequential technical amendments in which the public would have little or no interest. The VOR Federal airways, jet route, and compulsory reporting point listed in this document will be published as amended in §§ 71.123, 71.203, and 75.100 of Handbook 7400.7 effective November 1, 1991, which is incorporated by reference in 14 CFR 71.1.

The FAA has determined that this regulation only involves an established body of technical regulations for which frequent and routine amendments are necessary to keep them operationally current. It, therefore—(1) is not a "major rule" under Executive Order 12291; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034; February 26, 1979); and (3) does not warrant preparation of a regulatory evaluation as the anticipated impact is so minimal. Since this is a routine matter that will only affect air traffic procedures and air navigation, it is certified that this rule, when promulgated, will not have a significant economic impact on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 71

Aviation safety, VOR Federal airways, Compulsory reporting points, Jet routes, Incorporation by reference.

Adoption of the Amendment

In consideration of the foregoing, the Federal Aviation Administration amends 14 CFR part 71 as follows:

PART 71—[AMENDED]

1. The authority citation for 14 CFR part 71 continues to read as follows:

Authority: 49 U.S.C. App. 1348(a), 1354(a), 1510; E.O. 10854, 24 FR 9565, 3 CFR, 1959-1963 Comp., p. 389; 49 U.S.C. 106(g); 14 CFR 11.69.

§ 71.1 [Amended]

2. The incorporation by reference in 14 CFR 71.1 of the Federal Aviation Administration Order 7400.7, Compilation of Regulations, published April 30, 1991, and effective November 1, 1991, is amended as follows:

Section 71.123 Domestic VOR Federal Airways

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V-18

From Los Angeles, CA; Paradise, CA; Palm Springs, CA; Blythe, CA; Buckeye, AZ; Phoenix, AZ; INT Phoenix 155° and Stanfield, AZ, 105° radials; Tucson, AZ; Cochise, AZ; Columbus, NM; El Paso, TX; Salt Flat, TX; Wink, TX; Wink 060° and Big Spring, TX, 260° radials; Big Spring; Abilene, TX; Millsap, TX; Acton, TX; Scurry, TX; Quitman, TX; Texarkana, AR; Pine Bluff, AR; Holly Springs, MS; Jacks Creek, TN; Graham, TN; Nashville, TN; INT Nashville 102° and Hinch Mountain, TN, 285° radials; Hinch Mountain; Knoxville, TN; Holston Mountain, TN; Pulaski, VA; Roanoke, VA; Lynchburg, VA; Flat Rock, VA; Richmond, VA; INT Richmond 039° and Patuxent, MD, 228° radials; Patuxent; Smyrna, DE; Cedar Lake, NJ; Coyle, NJ; INT Coyle 036° and Kennedy, NY, 209° radials; Kennedy; Deer Park, NY; Calverton, NY; Norwich, CT; Boston, MA. The airspace within Mexico and the airspace below 2,000 feet MSL outside the United States is excluded. The airspace within Restricted Areas R-5002A, R-5002C and R-5002D is excluded during their times of use. The airspace within Restricted Areas R-4005 and R-4006 is excluded.

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V-29

From Snow Hill, MD; Salisbury, MD; Smyrna, DE; DUPONT, DE; Modena, PA; Pottstown, PA; East Texas, PA; Wilkes-Barre, PA; Binghamton, NY; INT Binghamton 005° and Syracuse, NY, 169° radials; Syracuse; Watertown, NY; INT Watertown 033° and Massena, NY, 241° radials; Massena.

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V-157

From Key West, FL; Miami, FL; INT Miami 337° and La Belle, FL, 124° radials, La Belle; Lakeland, FL; Ocala, FL; Gainesville, FL; Taylor, FL; Waycross, GA; Alma, GA; Allendale, SC; Vance, SC; Florence, SC. From Kinston, NC; Tar River, NC; Lawrenceville, VA; Richmond, VA; INT Richmond 039° and Patuxent, MD, 228° radials; Patuxent; Smyrna, DE; Woodstown, NJ; Robbinsville, NJ; INT Robbinsville 044° and LaGuardia, NY, 213° radials; LaGuardia; INT LaGuardia 032° and Deer Park, NY, 326° radials; INT Deer Park 326° and Kingston, NY, 191° radials; Kingston, NY; to Albany, NY. The airspace within R-2901A and R-6602A is excluded. The airspace at and above 7,000 feet MSL which lies within the Lake Placid MOA is excluded during the time the Lake Placid MOA is activated. The airspace within R-4005 and R-4006 is excluded.

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